Optimal control of slow-fast mechanical systems

Jean-Baptiste Caillau Univ. Côte d'Azur & CNRS/Inria

Joint work with

Lamberto Dell'Elce Inria (Sophia)

Jean-Baptiste Pomet Inria (Sophia)

Jérémy Rouot EPF (Troyes)

UCA Complex days Nice, January 2018



Minimum time.

$$\frac{\mathrm{d}I}{\mathrm{d}t} = \dot{I} = \varepsilon F_0(I, \varphi, \varepsilon) + \varepsilon \sum_{i=1}^m u_i F_i(I, \varphi, \varepsilon), \quad |u| = \sqrt{u_1^2 + \dots + u_m^2} \le 1,$$

$$\frac{\mathrm{d}\varphi}{\mathrm{d}t} = \dot{\varphi} = \omega(I) + \varepsilon G_0(I, \varphi, \varepsilon) + \varepsilon \sum_{i=1} u_i G_i(I, \varphi, \varepsilon),$$

with $I \in M$, $\varphi \in S^1$, $u \in \mathbb{R}^m$, fixed extremities I_0 , I_f , and free phases φ_0 , φ_f . Min. time Hamiltonian (Pontrjagin maximum principle):

$$\begin{split} H(I,\varphi,p_I,p_{\varphi},\varepsilon) &:= p_{\varphi}\omega(I) + \varepsilon K(I,\varphi,p_I,p_{\varphi},\varepsilon),\\ K &:= H_0 + \sqrt{\sum_{i=1}^m H_i^2},\\ H_i(I,\varphi,p_I,p_{\varphi},\varepsilon) &:= p_I F_i(I,\varphi,\varepsilon) + p_{\varphi} G_i(I,\varphi,\varepsilon), \quad i = 0, \dots, m. \end{split}$$

Averaged Hamiltonian. One defines

$$\overline{K} = \overline{H}_0 + \overline{K}_0, \quad \overline{H}_0 = \langle p_I, \overline{F}_0 \rangle,$$

$$\begin{split} \overline{K}_0(I,p_I) &:= \quad \frac{1}{2\pi} \int_0^{2\pi} \sqrt{\sum_{i=1}^m H_i^2(I,\varphi,p_I,p_{\varphi}=0,\varepsilon=0)} \,\mathrm{d}\varphi \\ &= \quad \frac{1}{2\pi} \int_0^{2\pi} \sqrt{\sum_{i=1}^m \langle p_I,F_i(I,\varphi,\varepsilon=0)\rangle^2} \,\mathrm{d}\varphi. \end{split}$$

Smooth on $\Omega:= \complement\overline{\Sigma}$ where

$$\begin{split} \boldsymbol{\Sigma} &:= \{ (I, p_I, \varphi) \in T^*M \times \mathbf{S}^1 \, | \, (\forall i = 1, m) : \langle p_I, F_i(I, \varphi, \varepsilon = 0) \rangle = 0 \}, \\ & \overline{\boldsymbol{\Sigma}} := \boldsymbol{\varpi}(\boldsymbol{\Sigma}) \, \, (\text{closed}), \quad \boldsymbol{\varpi} : T^*M \times \mathbf{S}^1 \to T^*M. \end{split}$$

One also defines the submanifold of M

$$M_0 := \Pi(\Omega)$$
 (open and assumed to be connex).

Assuming

(A1) rank{ $\partial^{j}F_{i}(I, \varphi, \varepsilon = 0)/\partial \varphi^{j}, i = 1, ..., m, j \ge 0$ } = n, $(I, \varphi) \in M \times S^{1}$, one has $\overline{K}_{0} : \Omega \subset T^{*}M \to \mathbb{R}$

$$\overline{K}_0(I,p_I) := \frac{1}{2\pi} \int_0^{2\pi} \sqrt{\sum_{i=1}^m \langle p_I, F_i(I,\varphi,\varepsilon=0) \rangle^2} \,\mathrm{d}\varphi,$$

positive definite (and 1-homogeneous): \overline{K}_0 defines a (symmetric) co-Finsler norm.

Remark. Condition related to the controllability of the original system without F_0 . (Lie brackets of F_1, \ldots, F_m with $\partial/\partial \varphi$.)

Finsler geometry

Finsler norm. $F: TM \to \mathbf{R}$ smooth on $TM \setminus 0$ s.t. (i) $F(x, \lambda v) = \lambda F(x, v), \ \lambda > 0$ (symmetric if F(x, -v) = F(x, v)) (ii) $\partial^2 F^2(x, v) / \partial v^2 > 0$ (tensor depending on v)

Metric. For x and y in M, set $d(x,y) := \inf t_f$ on all \mathscr{C}^1 curves γ s.t.

 $egin{aligned} &\gamma(0)=x, \quad \gamma(t_f)=y, \ &F(\gamma(t),\dot{\gamma}(t))\leq 1, \quad t\in [0,t_f]. \end{aligned}$

Geodesics. Constant speed curves whose short segments are length minimizing.

Finsler geometry

Co-Finsler norm. $F^*: T^*M \to \mathbf{R}$ smooth on $T^*M \setminus 0$ s.t. (i) $F^*(x,\lambda p) = \lambda F^*(x,p), \lambda > 0$ (ii) $\partial^2(F^*)^2(x,p)/\partial p^2 > 0$

Legendre transform. If F^* is a co-Finsler norm, then

$$F(x,v) := \max_{p \text{ s.t. } F^*(x,p) \leq 1} \langle p, v \rangle$$

defines a Finsler norm whose geodesics are integral curves of the Hamiltonian F^* restricted to the level $\{F^* = 1\}$. One has $F^*(x,p) = F(x,v)$, $v := \ell_x^*(p)$, where $\ell_x^* : T_x^*M \to (T_x^*M)^* \simeq T_xM$ is the Legendre transform

$$\ell_x^*: p \mapsto \frac{1}{2} \frac{\partial^2 (F^*)^2}{\partial p^2}(x,p)(p,\cdot)$$

$$\overline{K} = \overline{H}_0 + \overline{K}_0, \quad \overline{H}_0 = \langle p_I, \overline{F}_0 \rangle,$$

Assuming

(A2) $\overline{K}_0(I, \overline{F}_0^*(I)) < 1$, $I \in M_0$ (inverse Legendre transform of \overline{F}_0), one has $\overline{K} = \overline{H}_0 + \overline{K}_0$ positive definite: \overline{K} defines an asymmetric co-Finsler norm,

$$\overline{H}_{0} := \langle p_{I}, \overline{F}_{0} \rangle, \quad \overline{K}_{0}(I, p_{I}) := \frac{1}{2\pi} \int_{0}^{2\pi} \sqrt{\sum_{i=1}^{m} \langle p_{I}, F_{i}(I, \varphi, \varepsilon = 0) \rangle^{2} \, \mathrm{d}\varphi}$$

Remark. For small $\varepsilon > 0$, this condition is related to the local controllability of the original system, F_0 included. (Ability of F_1, \ldots, F_m and their brackets with $\partial/\partial \varphi$ to compensate for the drift F_0 .)

Controllability, existence, convergence.

Proposition. For $\varepsilon > 0$ small enough, controllability holds on M_0 .

(A3) The metric is geodesically convex on M_0 .

(A4) Compactness assumptions (related to singularities).

Proposition. Let I_0 and I_f belong to M_0 , $I_f \notin \text{Cut}(I_0)$. For $\varepsilon > 0$ small enough, there exist min. time trajectories connecting I_0 to I_f .

Theorem. Let I_0 and I_f belong to M_0 , $I_f \notin \operatorname{Cut}(I_0)$. Let $(I_{\varepsilon}, \varphi_{\varepsilon}, p_{I_{\varepsilon}}, p_{\varphi_{\varepsilon}})_{\varepsilon}$ be a family of minimizing extremals, and let $(\overline{t}_f(\varepsilon))_{\varepsilon}$ be the associated min. times. Then $(z_{\varepsilon} := (I_{\varepsilon}, p_{I_{\varepsilon}}))$,

$$\|z_{\varepsilon}-\overline{z}\|_{\infty} = O(\varepsilon) + O(k(\varepsilon)-1), \quad \varepsilon \overline{t}_{f}(\varepsilon) \to d(I_{0}, I_{f}), \quad \varepsilon \to 0,$$

where \overline{z} is the lift of the minimizing geodesic between I_0 and I_f .

Controlled Kepler equation. Work with the French Space Agency (CNES) and Thales Alenia Space. Due to super-integrability of the -1/|q| potential (Kepler), the min. time system

$$\ddot{q} = -\mu rac{q}{|q|^3} + rac{u}{M}, \quad |u| \leq T_{\max},$$

is a slow-fast control system with only one fast angle when restricted to transfer of a spacecraft between elliptical orbits. Averaging \implies Analysis of a symmetric Finsler metric in dimension five (no drift)

J2 perturbation. Due to the Earth oblateness, one has to take into account a small drift F_0 on the slow variables (equinoctial elements $I = (a, e, \omega, \Omega, i)$), defined by the J_2 potential $(1/|q|^3)$

$$R_0 = \frac{\mu J_2 r_e^2 \left(1 - e^2\right)^{-3/2}}{|q|^3} \left(\frac{1}{2} - \frac{3}{4}\sin^2 i + \frac{3}{4}\sin^2 i \cos(2\omega + 2 - \varphi)\right)$$

Reduction to one small parameter. The system has two small parameters (depending on the initial conditions I_0), one for the J_2 effect, one for the control:

$$\varepsilon_0 = \frac{3}{2} J_2 \frac{r_e^2}{a_0^2}, \quad \varepsilon_1 = \frac{a_0^2 T_{\text{max}}}{\mu M}$$

 r_e : Earth equatorial radius, a_0 : Initial semi-major axis, T_{max} : Maximum thrust level, M: Mass of the spacecraft Write

$$\begin{aligned} \frac{\mathrm{d}I}{\mathrm{d}t} &= \varepsilon_0 F_0(I,\varphi) + \varepsilon_1 \sum_{i=1}^m u_i F_i(I,\varphi), \\ &= \varepsilon \left(\lambda F_0(I,\varphi) + (1-\lambda) \sum_{i=1}^m u_i F_i(I,\varphi) \right), \end{aligned}$$

with $\varepsilon := \varepsilon_0 + \varepsilon_1$ and $\lambda := \varepsilon_0 / (\varepsilon_0 + \varepsilon_1)$.

Critical ratio. Two regimes depending on whether

(i) the J_2 effect is small compared to the control ($\varepsilon_0 \ll \varepsilon_1$ and $\lambda \to 0$), (ii) the J_2 effect dominates the control ($\varepsilon_0 \gg \varepsilon_1$ and $\lambda \to 1$). In terms of the averaged system,

$$\overline{K} = \lambda \overline{H}_0 + (1 - \lambda) \overline{K}_0$$

is a metric if and only if $\lambda < \lambda_c(I)$ with (*cf.* condition (A2))

$$\lambda_c(I) = \frac{1}{1 + \overline{K}_0(I, \overline{F}_0^*(I))}$$

 \implies Relevance of this critical ratio for the qualitative analysis of the original system?

Qualitative analysis on the min. time. Curve $\lambda \mapsto \tau_f(\lambda)$, $\tau_d \to 0$



Example : a = 30 Mm, e = 0.5, $\omega = \Omega = 0$, i = 51 deg (inclined medium orbit)



Qualitative analysis on the min. time. Curve $\lambda \mapsto \tau_f(\lambda), \ \tau_d \to 0$



Example : a = 30 Mm, e = 0.5, $\omega = \Omega = 0$, i = 51 deg (inclined medium orbit)

Conclusion and ongoing work

Well defined relevant limiting approximation: Asymmetric Finsler metric Caillau, J.-B.; Pomet, J.-B.; Rouot, J. Metric approximation of minimum time control systems. Preprint (2017).

▶ Complementary approaches (1/2): Filtering

Caillau, J.-B.; Dargent, T.; Nicolau, F. Approximation by filtering in optimal control and applications. *IFAC PapersOnLine* **50** (2017), no. 1, 1649–1654. Proceedings of the 20th IFAC world congress, Toulouse, July 2017.

▶ Complementary approaches (2/2): Higher order averaging

Dargent, T.; Pomet, J.-B.; Nicolau, F. Periodic averaging with a second order integral error. *IFAC PapersOnLine* **50** (2017), no. 1, 2892–2897. Proceedings of the 20th IFAC world congress, Toulouse, July 2017.

Ongoing work: Multi-phase averaging for min. time

Dell'Elce, L.; Caillau, J.-B.; Pomet, J.-B. Optimal low-thrust orbital transfer by averaging multiple frequencies. *7th International Meeting on Celestial Mechanics (CELMEC)*, San Martino, September 2017.

PEPS MSI (UCA) with Thales Alenia Space and Inria