

# Two-Frequency Averaging of Optimal Control Problems with Application to Time-Optimal Orbital Transfer



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## Low-thrust transfer: a recent problem in astrodynamics



# Fast motion of the satellite on a slowly-varying orbit



# Orbital perturbations may introduce new frequencies



Challenges:Do adjoint variables introduce additional fast dynamics?Is simple averaging enough when resonances are crossed?

# Outline



1. Dynamical systems with slow & fast dynamics

2. Averaging the two-phase optimal control problem



3. Near identity transformation of the initial state

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#### Averaging: an effective way to remove fast dynamics



f, g periodic in  $\phi, \epsilon \ll 1$  Slow variables:  $I \subset \mathbb{R}^n$  Fast variables:  $\phi \subset \mathbb{T}^r$ 

#### Is averaging compatible with resonance crossing?

$$\dot{I}_1 = \epsilon, \ \dot{I}_2 = \epsilon \cos(\phi_2 - \phi_1), \ \dot{\phi}_1 = I_1, \ \dot{\phi}_2 = I_2$$



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## Rich literature quantifying "small" and "most"



Results for 2 phases:

Arnold theorem very restrictive assumptions

**Neistadt theorem** restrictive assumptions, optimal estimate

P. LOCHAK, C. MEUNIER Multiphase Averaging for Classical Systems, Springer, 1988.

#### Optimal estimate for two-phase problem (Neistadt)

Strong assumption

$$\begin{aligned} \omega_2(\boldsymbol{J}) &\geq 0 \qquad \forall \, \boldsymbol{J} \in \boldsymbol{K} \\ \left| \frac{\mathsf{d}\omega}{\mathsf{d}t} \left( \boldsymbol{J}(\boldsymbol{J}_0, \boldsymbol{\phi}_0, t, \epsilon) \right) \right| &\geq c_1 \epsilon \qquad \forall \, \left( \boldsymbol{J}_0, \boldsymbol{\phi}_0 \right) \in \boldsymbol{K}' \times \mathbb{T}^2, \, t \leq \frac{1}{\epsilon} \qquad \text{where } \omega := \frac{\omega_1(\boldsymbol{J})}{\omega_2(\boldsymbol{J})} \end{aligned}$$

#### Main result

for 
$$\epsilon \to 0$$
,  $\exists K'' \subseteq K'$ ,  $\mu(K'' - K') < c\sqrt{\epsilon}$ , such that  $\forall (J_0, \phi_0, t) \in K'' \times \mathbb{T}^2$   
$$\sup_{t \in [0, \epsilon^{-1}]} ||I(t) - J(t)|| < c\sqrt{\epsilon} \log \frac{1}{\epsilon}$$



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#### Minimum time low-thrust transfer



What about the dynamics of the control?

#### Necessary conditions: are adjoints slow or fast?



In the single-phase problem,  $p_I$  is slow. See:

A. BOMBRUN, A.; J.B. POMET, The averaged control system of fast oscillating control systems SIAM J. Control Optim., 2013.

J.B. CAILLAU, J.B. POMET, J. ROUOT, J. CNES contract report, 2015.

## Adjoints remain slow in the multi-phase case

Main idea:

$$\omega(\mathbf{I}^*) \cdot \mathbf{p}_{\phi}^* = O(\epsilon) \quad \triangleright \quad \dot{\mathbf{p}}_{\mathbf{I}} = -\epsilon \frac{\partial K}{\partial \mathbf{I}} - \underbrace{\frac{\partial \omega}{\partial \mathbf{I}}}_{O(\epsilon)} \mathbf{p}_{\phi} = O(\epsilon)$$

Sketch of the proof:

Change of variables: 
$$I, \phi \to L, \begin{pmatrix} \alpha \\ \beta \end{pmatrix}$$
 s.t.  $L = I, \alpha = \frac{\omega \cdot \phi}{||\omega||}$  and  $\beta = \frac{\omega^{\perp} \phi}{||\omega||}$   
The Hamiltonian becomes:  $\mathcal{H} = ||\omega(L)||p_{\alpha} + \epsilon K (L, \alpha, \beta, p_L, p_{\alpha}, p_{\beta}, \epsilon)$   
By noting that  $p_{\alpha}(0) = p_{\beta}(0) = 0$  and normalizing  $||p_L(0)|| = 1$ , we have  $\mathcal{H} = \epsilon c$   
Hence,  $p_{\alpha}$  can be implicitly expressed as:  $p_{\alpha} = \epsilon \frac{c - K(L, \alpha, \beta, p_L, p_{\alpha}, p_{\beta}, \epsilon)}{||\omega(L)||}$   
Because  $\frac{\partial \mathcal{H}}{\partial p_{\alpha}} = ||\omega(L)|| + O(\epsilon) > 0$ , we have:  $p_{\alpha} = -\epsilon h(L, \alpha, \beta, p_L, p_{\beta}, \epsilon)$   
So that:  $\dot{p}_L = -\epsilon \frac{\partial H}{\partial p_{\alpha}} \frac{\partial h}{\partial L} = O(\epsilon)$ 

## Transfer to geostationary orbit



#### Averaged solution satisfies Neistadt's requirements



## Averaged solution plugged in the original system



#### Enhanced precision by averaging Moon gravity

