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Numerical Algorithms for Optimization and Control of PDE's Systems Application to Fluid Dynamics

R. Duvigneau

INRIA Sophia Antipolis-Méditerranée, OPALE Project-Team

Introduction

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Some examples and illustrations ...

- Global optimization of the wing shape of a business aircraft
- Local optimization of the shape of an airfoil
- Identification of optimal parameters for flow control

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Problem description (Piaggio-Aero P180)

- Drag minimization
- Lift constraint
- Global variables : span, root/tip chord ratio, sweep, twist and incidence angles
- Local variables : airfoil section (10 variables)
- Cruise regime $M_{\infty} = 0.83$
- Euler solver (Finite-Volume)
- Mesh 56,512 nodes 311,820 cells

Wing design



Wing design

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\rightarrow Convergence in about 80 iterations \sim 960 simulations

Wing design

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Comparison of initial (red) and final (blue) wing shapes

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Comparison of initial (top) and final (bottom) pressure fields

Wing design

Wing design

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Comparison of initial (top) and final (bottom) planforms

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Airfoil modification

Problem description

- Navier-Stokes, $k \omega$ turbulence modeling
- Reynolds number $Re = 19 \ 10^6$, Mach number 0.68
- Drag reduction



Airfoil modification

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\rightarrow Convergence in about 30 iterations \sim 150 simulations

Airfoil modification



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Problem description

- Navier-Stokes modeling (laminar flow)
- Reynolds number Re = 200
- Oscillatory rotating cylinder
- Objective : reduce the time-averaged drag
- Two control parameters : amplitude and frequency



Flow control

Flow control

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Comparison of vorticity fields without (top) and with control (bottom)

Flow control

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\rightarrow Convergence in about 5 iterations \sim 35 simulations

Flow control

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Surrogate model of drag coefficient

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How to carry out such an optimization process ?

- What are the tools involved ?
- How are they organized ?

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Optimization Algorithm

KIT German - French Summer School, September 2010

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Definition of an optimization problem

Minimize Subject to

nize	f(x)	$x \in \mathbb{R}^n$
ct to	$g_i(x) = 0$	$i=1,\cdots,I$
	$h_i(x) \ge 0$	$j=1,\cdots,m$

cost function equality constraints inequality constraints

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Definition of an optimization problem

Minimize	f(x)
Subject to	$g_i(x) =$
	h(x)

	f(x)	$x \in \mathbb{R}^n$
5	$g_i(x) = 0$	$i=1,\cdots,I$
	$h_i(x) \ge 0$	$j=1,\cdots,m$

cost function equality constraints inequality constraints

Role of the optimizer

Provide a candidate design vector $x \in \mathbb{R}^n$

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Definition of an optimization problem

Minimize	f(x)	$x \in \mathbb{R}^n$
Subject to	$g_i(x) = 0$	$i = 1, \cdots$
	$h_j(x) \geqslant 0$	$j=1,\cdots$

cost function equality constraints inequality constraints

Presentation

Role of the optimizer

Provide a candidate design vector $x \in \mathbb{R}^n$

Some approaches

 Descent methods : suitable for smooth convex functions steepest descent, conjugate gradient, quasi-Newton, Newton methods

. *m*

- Pattern search methods : suitable for noisy convex functions Nelder-Mead simplex, Torczon's multidirectional search algorithms
- Evolutionary methods : suitable for multimodal functions genetic algorithms, evolution strategies, particle swarm optimization, ant colony methods, simulated annealing

Steepest descent

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Algorithm

For each iteration k.

- Estimation of gradient $\nabla f(x_k)$
- Define search direction $d_k = -\nabla f(x_k)$
- Line search : find the step length ρ such as :
 - $f(x_k + \rho d_k) < f(x_k) + \alpha \nabla f(x_k) \cdot \rho d_k$ (Armijo rule) $f(x_k + \rho d_k) > f(x_k) + \beta \nabla f(x_k) \cdot \rho d_k$ (Goldstein rule)



Steepest descent

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Summary

Properties of steepest descent method:

- Proof of convergence to a local optimum
- Linear convergence rate :

$$\lim_{k \to \infty} \frac{\|x_{k+1} - x^*\|}{\|x_k - x^*\|} = a > 0$$

In practice:

- Unable to take into account function curvature
- Oscillatory optimization path

Quasi-Newton methods

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Algorithm

Improvement of the search direction:

Definition according to the function curvature:

$$\widetilde{H}_k d_k = -\nabla f(x_k),$$

with \tilde{H} approximation of the Hessian matrix (second-order derivative) **Iterative** construction of the Hessian matrix (BFGS formula) :

$$\begin{split} \widetilde{H}_0 &= Id \\ \widetilde{H}_{k+1} &= \widetilde{H}_k - \frac{1}{s_k^T \widetilde{H}_k s_k} \widetilde{H}_k s_k s_k^T \widetilde{H}_k^T + \frac{1}{y_k^T s_k} y_k y_k^T, \end{split}$$
with $s_k = x_{k+1} - x_k$ and $y_k = \nabla f(x_{k+1}) - \nabla f(x_k)$

Quasi-Newton methods

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Summary

Properties of quasi-Newton methods:

Take into account the curvature of the past iteration:

$$\widetilde{H}_{k+1}(x_{k+1}-x_k) =
abla f(x_{k+1}) -
abla f(x_k)$$

- H_k positive definite (if line search efficient enough)
- Proof of convergence to local optimum
- Super-linear convergence rate :

$$\lim_{k \to \infty} \frac{\|x_{k+1} - x^*\|}{\|x_k - x^*\|} = 0$$

In practice:

- Very efficient algorithm for convex problems (close to Newton method)
- Gradient should be available

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(λ, μ) -Evolution Strategy

Algorithm

For each iteration k, according to \bar{x}_k and $\bar{\sigma}_k$:

- Generate a set of λ perturbation lengths $\sigma_i = \bar{\sigma}_k e^{\tau N(0,1)}$
- Generate a sample of λ individuals $x_i = \bar{x}_k + \sigma_i N(0, Id)$ (mutation) where N(0, Id) multivariate normal distribution with 0 mean and Id covariance matrix
- Evaluate the performance of λ individuals
- Choose the best μ parents among λ individuals (selection)
- Update distribution properties (crossover and self-adaption) :

$$ar{\mathbf{x}}_{k+1} = rac{1}{\mu} \sum_{i=1}^{\mu} \mathbf{x}^i \qquad ar{\sigma}_{k+1} = rac{1}{\mu} \sum_{i=1}^{\mu} \sigma^i$$



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(λ, μ) -Evolution Strategy

Summary

Properties of evolution strategies:

Proof of convergence to the global optimum in a statistical sense

e.g. $\forall \epsilon > 0$ $\lim_{k \to \infty} P(|f(x_k) - f(x^*)| \leq \epsilon) = 1$

linear convergence rate

In practice:

- Able to avoid local optima
- Curse of dimensionality
- Low local convergence rate due to isotropic search



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Covariance Matrix Adaption (CMA-ES)

Algorithm

Improvement by using an anisotropic distribution:

• Use of a covariance matrix C_k to generate the population:

$$\mathbf{x}_i = \bar{\mathbf{x}}_k + \bar{\sigma}_k \ \mathbf{N}(0, C_k)$$

 $= \bar{\mathbf{x}}_k + \bar{\sigma}_k \ \mathbf{B}_k \mathbf{D}_k \mathbf{N}(0, Id)$

with B_k matrix of eigenvectors of $C_k^{1/2}$ and D_k diagonal matrix of eigenvalues

Iterative construction of the covariance matrix :

 $C_0 = Id$

$$C_{k+1} = \underbrace{(1-c)C_k}_{\text{previous estimate}} + \underbrace{\frac{c}{m}p_kp_k^T}_{\text{1D update}} + \underbrace{c(1-\frac{1}{m})\sum_{i=1}^{\mu}\omega^i(y^i)(y^i)^T}_{\text{update}}$$

covariance of parents

with :

p_k evolution path (moves performed during last iterations)
 yⁱ = (*xⁱ* − *x̄_k*)/*σ_k*

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Covariance Matrix Adaption (CMA-ES)

Summary

Properties:

• C_k is an approximation of the inverse of the Hessian matrix



- Several invariance properties:
 - Invariance under order-preserving transformations of the function
 - Invariance under scaling of the search space
 - Invariance under rotation of the search space

In practice:

- Outperforms most evolutionary methods
- Only a few parameters defined by the user

Analytical example

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Rosenbrock function

- Non-convex function "Banana valley"
- Dimension n = 16




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quasi-Newton

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Camel back function

- Dimension n = 2
- Six local optima
- Two global optima

camel back : f(x,y)=(4-2.1x^2+x^4/3)x^2 +xy + 4(-1+y^2)y^2+1.0317



solution point 🛛 🕷



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Optimization path

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Optimization path



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Optimization path

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Role of the parameterization tool

- Describe the system with a few variables
- Transform a given design vector to a new engineering system

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Role of the parameterization tool

- Describe the system with a few variables
- Transform a given design vector to a new engineering system

A particular case: shape parameterization

- Discrete shape : set of moving nodes
- Parametric surface : Bézier, B-Splines, NURBS with control points
- Composite approach : set of baseline components with parameters
- Free-form deformation : deform a lattice that embeds the system

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Discrete shape representation

Principle

- Shape described by a surface grid
- Each node can be moved independently



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Discrete shape representation

Principle

- Shape described by a surface grid
- Each node can be moved independently



Advantages

- No strong assumption concerning the search space
- No need for CAD software (CAD-free)

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Principle

- Shape described by a surface grid
- Each node can be moved independently



Advantages

- No strong assumption concerning the search space
- No need for CAD software (CAD-free)

Drawbacks

- Smoothing required at each optimization step
- Very large number of design variables (\sim 10,000)
- Optimized shape = surface grid

Discrete shape representation

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Conclusion

Shape described by a parametric surface: $x(\xi) = \sum_{i=0}^{n} N_i(\xi) X_i$

Choice of basis functions N_i:

Principle

- Bézier : global influence, C^{∞}
- B-Spline : compact support, C^{k-2} (k order)
- NURBS : conic surfaces possible



Parametric shape representation

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Principle

Shape described by a parametric surface: $x(\xi) = \sum_{i=0}^{n} N_i(\xi) X_i$

- Choice of basis functions N_i:
 - Bézier : global influence, C^{∞}
 - B-Spline : compact support, C^{k-2} (k order)
 - NURBS : conic surfaces possible



Advantages

- Low number of design variables (\sim 10)
- Optimal shape is parametric and smooth
- Hierarchical representation

Parametric shape representation

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Principle

Shape described by a parametric surface: $x(\xi) = \sum_{i=0}^{n} N_i(\xi) X_i$

- Choice of basis functions N_i:
 - Bézier : global influence, C^{∞}
 - B-Spline : compact support, C^{k-2} (k order)
 - NURBS : conic surfaces possible



Advantages

- Low number of design variables (\sim 10)
- Optimal shape is parametric and smooth
- Hierarchical representation

Drawbacks

Mild assumption concerning the design space

Parametric shape representation

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Composite shape representation

Principle

 Use a set of baseline components (edges, circles, etc) with parameters (lengths, angles, positions)

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Principle

 Use a set of baseline components (edges, circles, etc) with parameters (lengths, angles, positions)

Advantages

- Optimal shape obtained from baseline components (manufacturing constraint)
- Low number of design variables
- Variables chosen according to engineering experience

Composite shape representation

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Principle

 Use a set of baseline components (edges, circles, etc) with parameters (lengths, angles, positions)

Advantages

- Optimal shape obtained from baseline components (manufacturing constraint)
- Low number of design variables
- Variables chosen according to engineering experience

Drawbacks

Strong assumption concerning the design space

Composite shape representation

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Principle

- Represent deformation instead of the shape
- Embed the the system in a lattice, described as parametric volume





Free-Form Deformation (FFD)

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Principle

- Represent deformation instead of the shape
- Embed the the system in a lattice, described as parametric volume





Free-Form Deformation (FFD)

Advantages

Low number of design variables, whatever the system complexity

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Principle

- Represent deformation instead of the shape
- Embed the the system in a lattice, described as parametric volume



Advantages

Low number of design variables, whatever the system complexity

Drawbacks

- Not easy to handle
- No representation of the optimal shape

Free-Form Deformation (FFD)

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Role of the grid generation tool

- Generate a grid in accordance with the parameterized geometry
- Requirement: suitable for computations

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Role of the grid generation tool

- Generate a grid in accordance with the parameterized geometry
- Requirement: suitable for computations

Difficulties

- Automated task
- Need for robustness (large variations of geometry)
- Need for accuracy (mesh quality maintained)

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Role of the grid generation tool

- Generate a grid in accordance with the parameterized geometry
- Requirement: suitable for computations

Difficulties

- Automated task
- Need for robustness (large variations of geometry)
- Need for accuracy (mesh quality maintained)

Possible Approaches

- Generate completely a new grid
- Deform an existing reference grid

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Complete mesh generation

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Principle

- Store all the steps of the mesh construction (usually in a script) with some parameters as input (geometry)
- Run the grid generation software automatically for each new geometry

Complete mesh generation

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Principle

- Store all the steps of the mesh construction (usually in a script) with some parameters as input (geometry)
- Run the grid generation software automatically for each new geometry

Advantages

Very large geometry changes are possible

Complete mesh generation

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Principle

- Store all the steps of the mesh construction (usually in a script) with some parameters as input (geometry)
- Run the grid generation software automatically for each new geometry

Advantages

Very large geometry changes are possible

Drawbacks

- Regularity of the mesh w.r.t. geometry change ?
- treatment of boundary layers tedious

Illustration

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Reference mesh deformation

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Principle

- Construct a reference mesh (initial geometry)
- Move the nodes according to geometry change (same topology)
- Need to solve an extra problem for the displacement field (e.g. structural analogy)

Reference mesh deformation

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Principle

- Construct a reference mesh (initial geometry)
- Move the nodes according to geometry change (same topology)
- Need to solve an extra problem for the displacement field (e.g. structural analogy)

Advantages

- Control of the grid quality
- Smoothness of the deformation

Reference mesh deformation

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Principle

- Construct a reference mesh (initial geometry)
- Move the nodes according to geometry change (same topology)
- Need to solve an extra problem for the displacement field (e.g. structural analogy)

Advantages

- Control of the grid quality
- Smoothness of the deformation

Drawbacks

Moderate geometry change

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Example: discrete mechanical analogy

Method

- Consider all edges as lineal springs and all angles as torsional springs
- Solve a linear system representing the mechanical equilibrium:

$$(K_{lin} + K_{tors} | Id) q = \begin{pmatrix} 0 \\ \overline{q} \end{pmatrix}$$

with: *K* stiffness matrices *q* displacement

 \overline{q} imposed boundary displacement



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Example: discrete mechanical analogy



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Example: use of Free-Form Deformation


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Role of sensitivity analysis

• Compute the gradient of the cost function (e.g. drag) with respect to design variables

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Role of sensitivity analysis

• Compute the gradient of the cost function (e.g. drag) with respect to design variables

Possible Approaches

- Finite-difference approximation
- Complex estimation
- Adjoint approach

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Finite-difference approximation of the gradient

Method

Perform n + 1 or 2n simulations for perturbed design variable values x ± ee_i
 Compute approximated gradient:

$$|\nabla f(x)|_i \simeq rac{f(x+\epsilon e_i)-f(x)}{\epsilon}$$
 $|\nabla f(x)|_i \simeq rac{f(x+\epsilon e_i)-f(x-\epsilon e_i)}{2\epsilon}$

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Finite-difference approximation of the gradient

Method

Perform n + 1 or 2n simulations for perturbed design variable values x ± ee_i
 Compute approximated gradient:

$$|\nabla f(x)|_i \simeq rac{f(x+\epsilon e_i)-f(x)}{\epsilon} \qquad |\nabla f(x)|_i \simeq rac{f(x+\epsilon e_i)-f(x-\epsilon e_i)}{2\epsilon}$$

Drawbacks

- Choice of perturbation coefficient ϵ
- \blacksquare Low accuracy \rightarrow noisy cost function
- Very expensive for large n

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Finite-difference approximation of the gradient

Method

Perform n + 1 or 2n simulations for perturbed design variable values x ± ee_i
 Compute approximated gradient:

$$|
abla f(x)|_i \simeq rac{f(x+\epsilon e_i)-f(x)}{\epsilon} \qquad
abla f(x)|_i \simeq rac{f(x+\epsilon e_i)-f(x-\epsilon e_i)}{2\epsilon}$$

Drawbacks

- Choice of perturbation coefficient ϵ
- Low accuracy \rightarrow noisy cost function
- Very expensive for large n

Advantages

Non-intrusive approach

Complex estimation

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• Move all real variables x of the code to complex variables: z = x + iy

- The cost function (output) is also complex: f = u + iv
- The Cauchy-Riemann condition : $\frac{\partial u}{\partial x} = \frac{\partial v}{\partial y} \simeq \frac{v(x+i(y+h))-v(x+iy)}{h}$
- But values are real : y = 0 v(x) = 0 and u = f
- Then, the following approximation can be used:

$$abla f(x) \simeq rac{Im[f(x+ih)]}{h}$$

Propagate imaginary part of input variable to estimate gradient

Complex estimation

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Accurate gradient estimation (no difference)

Complex estimation

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Propagate imaginary part of input variable to estimate gradient

Advantages

Method

Accurate gradient estimation (no difference)

Drawbacks

- Move CFD code to complex !
- n simulations required

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Let consider the cost function as a constrained functional:

$$f: x \mapsto f(x) = F(x, W) \in \mathbb{R}$$

with $W = W(x) \in \mathbb{R}^N$ is the solution of the state equation:

$$\Psi(x,W)=0$$

Apply chain rule:

$$\frac{df}{dx_i} = \frac{\partial F}{\partial x_i} + \frac{\partial F}{\partial W} \frac{dW}{dx_i}$$
$$\frac{\partial \Psi}{\partial x_i} + \frac{\partial \Psi}{\partial W} \frac{dW}{dx_i} = 0$$

by combining equations:

$$\frac{df}{dx_i} = \frac{\partial F}{\partial x_i} - \frac{\partial F}{\partial W} \left(\frac{\partial \Psi}{\partial W}\right)^{-1} \frac{\partial \Psi}{\partial x_i}$$

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Finally, gradient can be computed by solving the (linear) system first:

$$\left(\frac{\partial \Psi}{\partial W}\right)^T \Pi = \left(\frac{\partial F}{\partial W}\right)^T$$

where Π are the adjoint variables, and then by computing:

$$\left(\frac{df}{dx}\right)^T = \left(\frac{\partial F}{\partial x}\right)^T - \left(\frac{\partial \Psi}{\partial x}\right)^T \Pi$$

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Advantages

Method

- Adjoint system do not depend on x
- Cost rather independent from the size of x

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$$\left(\frac{df}{dx}\right)^{T} = \left(\frac{\partial F}{\partial x}\right)^{T} - \left(\frac{\partial \Psi}{\partial x}\right)^{T} \Pi$$

Advantages

- Adjoint system do not depend on x
- Cost rather independent from the size of x

Drawbacks

The adjoint system has to be constructed (intrusive approach)

Inversion of adjoint system difficult (badly conditionned)

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Some reasons to use surrogate models

- Cost function evaluations are very expensive
- Some results are just discarded (in particular with ES)
- Weak use of past results (iterative process without history)

Presentation

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Some reasons to use surrogate models

- Cost function evaluations are very expensive
- Some results are just discarded (in particular with ES)
- Weak use of past results (iterative process without history)

Principles

- Store all cost function values into a database
- Use the database to build a surrogate model
- Use the surrogate model as cheap and approximate evaluation

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Principles

- Store all cost function values into a database
- Use the database to build a surrogate model
- Use the surrogate model as cheap and approximate evaluation

Some possible surrogate models

- Radial basis functions (RBFs)
- Artificial Neural Networks (ANNs)
- Gaussian processes (kriging)

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Radial Basis Functions (RBF)

Evaluation

Approximation of the function $\mathcal{J}(\mathbf{x})$, $\mathbf{x} \in \Re^n$ of the form:

$$f(\mathbf{x}) = \sum_{j=1}^{N_c} \omega_j \ \phi_j(\mathbf{x}) \tag{1}$$

where ϕ_i are radial functions:

$$\phi_j(\mathbf{x}) = \Phi(\|\mathbf{x} - \mathbf{x}_j\|) \quad \Phi(r) = e^{-\frac{r^2}{s^2}}$$
(2)

 $(\mathbf{x}_j)_{j=1,...,N_c}$ s $(\omega_j)_{j=1,...,N_c}$ points stored in the database attenuation factor weights adjusted to fit the data

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Radial Basis Functions (RBF)

Training

 $(\omega_i)_{i=1,\ldots,N_c}$ are determined from interpolation conditions:

$$f(\mathbf{x}_i) = \sum_{j=1}^{N_c} \omega_j \phi_j(\mathbf{x}_i) \quad i = 1, \dots, N_c$$
(3)

 $(\omega_i)_{i=1,\ldots,N_c}$ is the solution of the linear system:

$$\begin{pmatrix} \phi_{1}(\mathbf{x}_{1}) & \dots & \phi_{N_{c}}(\mathbf{x}_{1}) \\ \phi_{1}(\mathbf{x}_{2}) & \dots & \phi_{N_{c}}(\mathbf{x}_{2}) \\ \dots & \dots & \dots & \dots \\ \phi_{1}(\mathbf{x}_{N_{c}}) & \dots & \phi_{N_{c}}(\mathbf{x}_{N_{c}}) \end{pmatrix} \begin{pmatrix} \omega_{1} \\ \omega_{2} \\ \dots \\ \omega_{N_{c}} \end{pmatrix} = \begin{pmatrix} \mathcal{J}(\mathbf{x}_{1}) \\ \mathcal{J}(\mathbf{x}_{2}) \\ \dots \\ \mathcal{J}(\mathbf{x}_{N_{c}}) \end{pmatrix}$$
(4)

s set by the user or optimized by internal algorithm

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Principle

• The vector of known function values F_N is assumed to be one realization of a Gaussian process:

$$p(F_N) = \frac{\exp\left(-\frac{1}{2}F_N^\top C_N^{-1}F_N\right)}{\sqrt{(2\pi)^N \det(C_N)}}$$

with a given covariance matrix $C_{mn} = c(x_m, x_n)$.

It can be shown that (conditional probabilities):

$$p(f_{N+1}|F_N) \propto \exp\left[-rac{(f_{N+1} - \hat{f}_{N+1})^2}{2\sigma_{f_{N+1}}^2}
ight]$$

where:

$$\hat{f}_{N+1} = k^\top C_N^{-1} F_N, \qquad \sigma_{f_{N+1}}^2 = \kappa - k^\top C_N^{-1} k$$

Gaussian processes (Kriging)

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Illustration for 1D problem

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Gaussian Processes (Kriging)

Choice of the covariance function

Distance dependent correlation, scaling, offset

$$c(x,y) = heta_1 \exp\left[-rac{1}{2}\sum_{i=1}^d rac{(x_i-y_i)^2}{r_i^2}
ight] + heta_2,$$

Parameters to be optimized $\Theta = (\theta_1, \theta_2, r_1, r_2, \dots, r_d)$

Choice of the parameters (training phase)

- Choose Θ to maximize the likelihood of the known function values
- Internal optimization

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Application to Inexact Pre-Evaluation (IPE)

Principle

- For evolutionary optimizers several evaluations are just not used
- Use surrogate models as pre-screening criterion for CFD evaluation

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Application to Inexact Pre-Evaluation (IPE)

Principle

- For evolutionary optimizers several evaluations are just not used
- Use surrogate models as pre-screening criterion for CFD evaluation

Expected benefits

- Avoid useless evaluations for evolutionary optimizers
- Reduce drastically the cost of evolutionary optmizers
- Low coupling between optimizer and metamodel

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Inexact Pre-Evaluation (IPE) approach

- **1** Set n = 0
- 2 If $n \leq N_e$ compute cost function $f(x_k^n), k = 1, ..., K$ using the exact model, else using metamodel $\tilde{f}(x_k^n), k = 1, ..., K$.
- **3** If $n > N_e$, then select a subset of points S^n for exact evaluation.
- Update the optimizer parameters (mean, standard deviation) using only the exactly evaluated cost functions
- 5 Store exactly evaluated function values into a database
- **6** If $n < N_{\text{max}}$, then n = n + 1 and go to step (iii), else STOP.

Illustration for business jet

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 CFD evaluations for the 10% best points or points with predicted improvement

- 500 iterations
- RBF model with local database (40 pts)

Number of exact evaluations



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Number of exact evaluations

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Application to Efficient Global Optimization (EGO)

Main ideas

- Use a metamodel to drive the search
- Use a Gaussian Process model to take into account the probability of obtaining a better design

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Application to Efficient Global Optimization (EGO)

Main ideas

- Use a metamodel to drive the search
- Use a Gaussian Process model to take into account the probability of obtaining a better design

Expected benefits

- Global optimization (proof of convergence)
- Deterministic approach
- Measure of the probability of improvement

EGO algorithm

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Basics

- Build iteratively a database and corresponding Gaussian process
- Enrichement chosen in order to:
 - Minimize the current Gaussian process model
 - Explore where the probability of improvement is high

EGO algorithm

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Basics

- Build iteratively a database and corresponding Gaussian process
- Enrichement chosen in order to:
 - Minimize the current Gaussian process model
 - Explore where the probability of improvement is high

Algorithm

- Build an a priori database (Latin Hypercube Sampling)
- Construct a global Gaussian process
- **3** Find the points x_i^* that minimize / maximize a merit function :
 - Statistical lower bound
 - Probability of improvement
 - Expected improvement
- **4** Evaluate the p points $(x_i^{\star})_{i=1,\ldots,p}$ and add them in the database
- 5 Return to step 2 until convergence

Illustration for business jet

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EGO algorithm with three merit functions

- 150 iterations
- Initial database size : 60 points

Cost function



Comparison IPE - EGO

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0.6

0.5 L

1000

2000

3000

Number of CFD

4000

5000

6000

7000

Use of surrogate models

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Main ideas

- Use all evaluations already performed
- Use expensive simulation only if required

Use of surrogate models

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Main ideas

- Use all evaluations already performed
- Use expensive simulation only if required

Advantages

- Drastic reduction of CPU cost
- Interesting for post-processing / interactive study

Use of surrogate models

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Main ideas

- Use all evaluations already performed
- Use expensive simulation only if required

Advantages

- Drastic reduction of CPU cost
- Interesting for post-processing / interactive study

Drawbacks

- More sophisticated approach
- Restricted to problems of low dimension
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Optimization and control in CFD:

- A multi-disciplinary field:
 - Applied mathematics
 - Numerical methods
 - Physical phenomena
 - Geometry
 - Computer sciences
- Efficiency of methods are strongly problem dependent
- Efficiency depends strongly of the global coherency of the loop

Conclusion

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Several topics have not been discussed:

- Multi-objective optimization: How to minimise several criteria ?
- Multi-disciplinary optimization: How to couple several disciplines ?
- Constrained optimization: How to take into account constraints ?
- Robust optimization: How to take into account uncertainties ?
- Hierarchical optimization: How to develop multi-level strategies ?
- Distributed optimization: How to use parallel computing ?
- Automatic differentiation: How to use AD softwares ?
- ...